

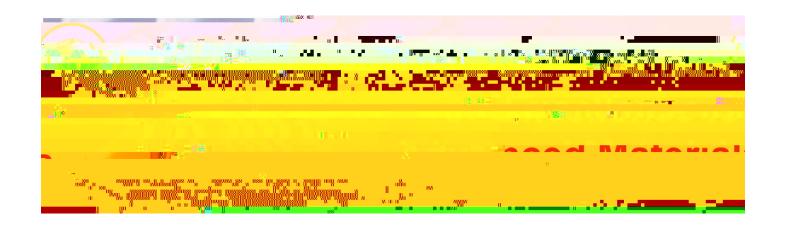
Factors Affecting Qualification/Certification - Evaluating the Criticality of Inherent Anomalies/Defects on the Fatigue Behavior of Additively Manufactured Ti-6Al-4V Parts

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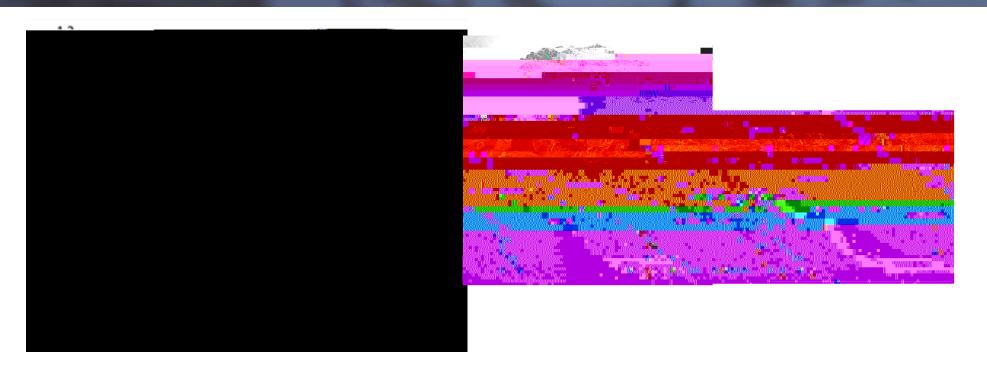
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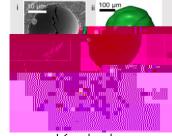


AM defects:

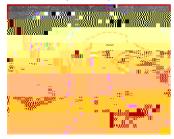
Significantly reduce and introduce uncertainty to fatigue performance Pose great challenge for qualification/certification of AM parts



Gas-entrapped pores (GEPs)



Keyholes (KHs)



Lack of fusions (LoFs)

: To quantify the detrimental effect of volumetric defects on mechanical properties of L-PBF Ti-6AI-4V

Gr. 5

: Three steps are taken,

- Lexplore process windows by varying laser power, scan speed, and hatching distance
- Determine the criticality of volumetric defects on mechanical performance using specimens seeded with different defect types
- III. Take advantage of machine learning and simulations wherever applicable



AP&C Ti-6Al-4V Grade 5 powder (15-53 µm) was used as feedstock

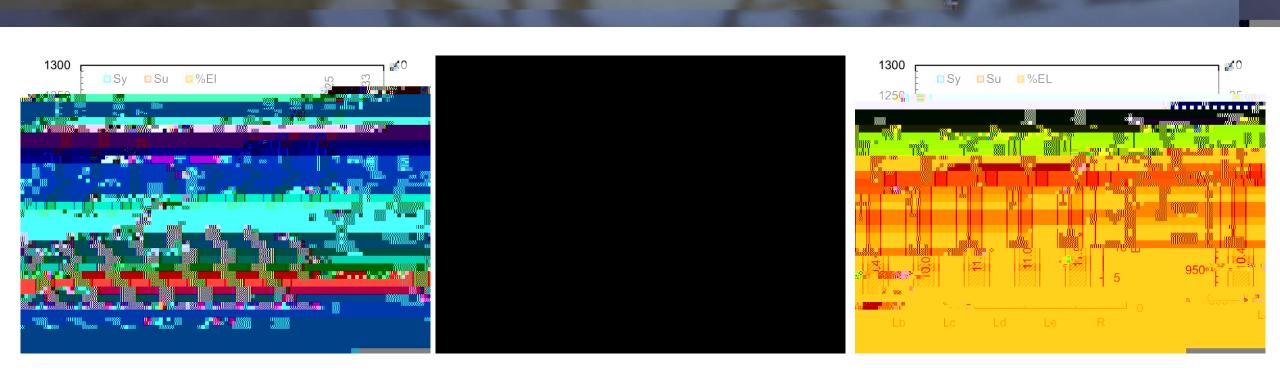
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P: Laser power
V: Laser speed
h: Hatch distance
t: Layer thickness

Note: Original plan was to fabricate specimens with 6 sets of process parameters, we fabricated 2 two extra sets (Ld and Le) X-ray computed tomography (XCT) was performed on vertical fatigue specimens with 5.5 µm voxel size

240 fatigue (16 x 15) and 96 tensile (16 x 6) specimens were fabricated Lack of fusion (LoF): P^{-5%}, P^{-10%}, P^{-20%}, H^{+5%}, and H^{+20%} Keyhole (KH): P^{+30%}V^{-20%} and P^{+20%}V^{-30%}

KH specimens were fabricated only in vertical orientation, while the recommended (R) and LoF ones were fabricated in vertical, diagonal, and horizontal orientations



Yield strength (YS) and ultimate tensile strength (UTS) of all specimens were almost comparable KH specimens had slightly higherslightly



Fatigue specimens were tested all the way until final fracture

In vertical orientation, KH specimens exhibited better fatigue performance than recommended ones

Fatigue lives of LoF specimens had more scatter than KH ones due to wide variation in shape, size, and location of the crack initiating defects

LoF specimens with higher defect content (Sets La, Lb, and Lc) exhibited worse fatigue performance for vertical and diagona 5 400 gestion worse ecommended ones

Defect sizes were measured using actual ¿a ea of the defect

The size of the fatigue crack initiating defects of recommended and KH



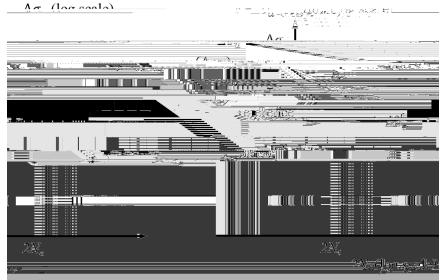


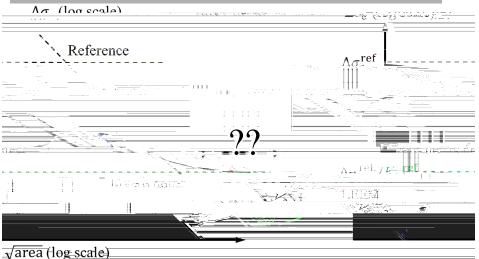
Defects are inherent to the AM process and their morpholoe96 o641 rg2 T14305 0 960 540 reW*n4 reW*n119.16 0 0



Simultaneous usage of several discriminating parameters can more effectively describe the geometric feature of a defect

Although different defect types have overlaps in all parameters, usage of more parameters can improve the classification accuracy



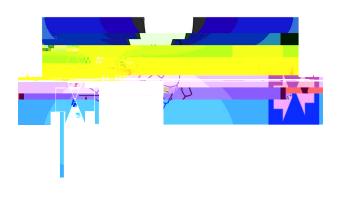


Volumetric/surface defects act as stress risers and can accelerate fatigue crack initiation

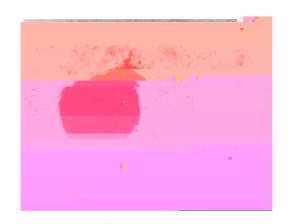
Notch-factor approach can account for the effect of volumetric/surface defects

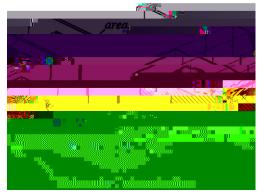
The extremely large defects behave like macro notches.

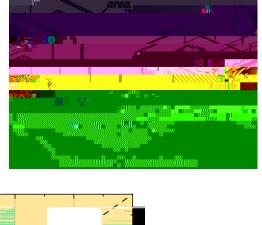
How to calculate the effect of defects of different sizes and shapes?

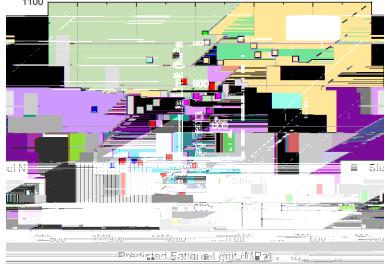


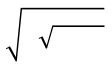
akamiÑ defec en i i e fa ig e (DSF) model



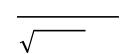




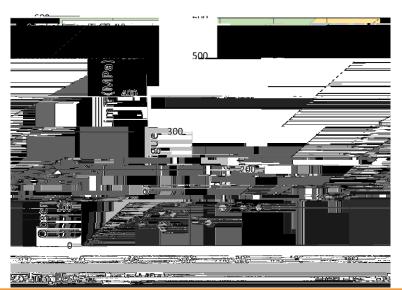




Surface defects: Internal defects:



Surface defects: Internal defects:



AM materials loaded below fatigue limit are believed to endure indefinitely but are

Instead, small cracks initiated from defects can be arrested

The rapidly reducing stress away from defects can cause the effective stress intensity factor range () to first decrease before increasing:

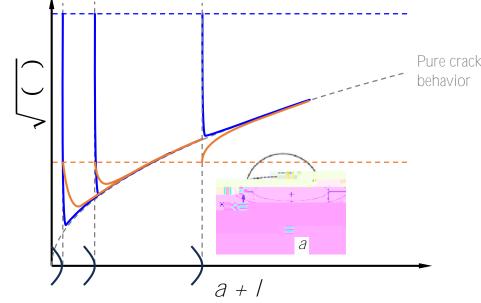
$$() \quad \sqrt{()}$$

The fatigue limit of a defect-laden material can be calculated:

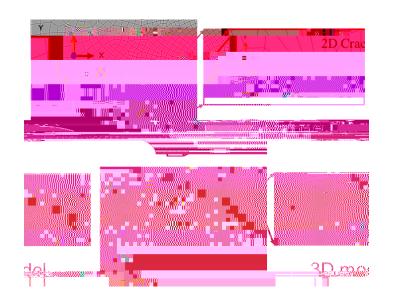
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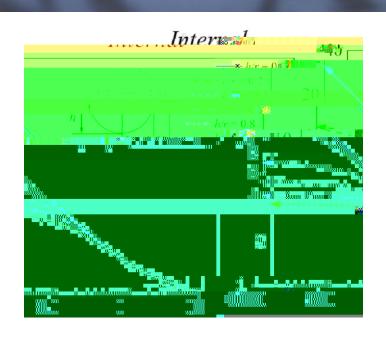
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How to calculate ()?









3D LE-FEA is performed to calculate stress intensity factor of cracks from internal and surface defects. The faige life piedic ion imade by helic ack a expanding it are all factors.

How about surface defects?

KH specimens exhibited better fatigue performance than recommended ones due to smaller crack initiating defect sizes

LoF specimens exhibited more scatter in fatigue life due to differences in crack initiating defect sizes and their morphologies

A morphological parameter alone could not explain the scatter in fatigue life

Larger defects, located at or near surface, were more

National Center for Additive Manufacturing Excellence (NCAME)

